

(NASA-CR-143888) SYSTEM TEST REPORT: LST SECONDARY MIRROR ARTICULATION MECHANISM P/N 660-0417 Final Report, 1 Jan. - 30 Nov. 1974 (Perkin-Elmer Corp.) 38 p CSCL 20K N75-27393

Unclas G3/37 29186



PERKIN-ELMER

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# ELECTRO-OPTICAL DIVISION NORWALK, CONNECTICUT

REPORT NO. 12507

SYSTEM TEST REPORT - LST SECONDARY

MIRROR ARTICULATION MECHANISM

P/N 660 - 0417

### PREPARED FOR

NATIONAL AERONAUTICS AND SPACE

ADMINISTRATION, MARSHALL SPACE

FLIGHT CENTER, HUNTSVILLE, ALABAMA

Contract No. NAS 8-29723

DATE: May 10, 1975

Prepared By:

W. E. Kohman

Approved By:

D. J. McCarthy

### SYSTEM TEST REPORT - LST SECONDARY MIRROR ARTICULATION SYSTEM

### 1.0 INTRODUCTION

This report documents work performed under Contract NAS 8-29723

Modification No. 2. The activity relates to LST Secondary Mirror Actuation

System 660-0267. The system was modified as defined by 660-0417 and re-tested.

#### 2.0 SUMMARY

The modifications performed to convert the system to the 660-0417 configuration are listed as follows:

- 2.1 Reduce the dummy mirror size to 40cm diameter. The mass corresponds to the LST secondary mirror.
- 2.2 Remove the two degree of freedom alignment drive mechanism.
- 2.3 Assemble the modified mirror mass, mirror support flexure, piezoceramic actuators, and position transducers to a simple circular plate so as to provide only the image motion compensation function.
- 2.4 Perform a simple test series to assure operation comparable to that achieved with the original larger sized assembly.

### 3.0 DESCRIPTION OF SYSTEM MODIFICATION

Details of the modification are shown on drawing 660-0417. The calculated weight and inertia relationships are as follows.

	LST Secondary Mirror	Mass <u>Simulator</u>
Clear Aperture	38.5cm	-
Diameter	40.0cm	40cm
Weight	23.1 <sub>1</sub> 1b.	22.6 lb.
Inertia	.961 in-lb-sec <sup>2</sup>	.924 in-1b-sec <sup>2</sup>

### 4.0 SYSTEM TEST

The system was tested in accordance with selected sections of Test Procedure ER 11783. The position sensors were re-calibrated and the actuator voltage to position gain was determined for each axis. The frequency response and threshold were determined for each axis. Data sheets, calculations and plotted data are attached.

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CHKD. BY DATE LOT SUBJECT TEST PROCEDURE SHEET NO. OF TO CHKD. BY DATE LOT SECONDARY MIRROR JOB NO.

I DETERMINE VOLTAGE ANGLE GAIN-ELEVATION AXIS

2 MIRROR AZ = 0

5. AVERAGE MONITOR GAIN

土 2.18 %、

6. SCALE FACTOR - MIRROR ELEVATION

BY	DATE	SUBJECT	SHEET NOOF
CHIER DV	DATE		100 110

I 4.2.2.1 TOTAL ANGULAR RANGE

$$III 4,2,2,2$$
 FREQUENCY RESPONSE @ 4 HZ
$$db = 20 lob I_{0} = 20 lob 465$$

$$I_{0} = 20 lob 700$$

BY\_\_\_\_\_\_ DATE\_\_\_\_\_\_ SUBJECT\_\_\_\_\_\_\_ SHEET NO. 3 OF 7
CHKD. BY\_\_\_\_ DATE\_\_\_\_\_\_ JOB NO. \_\_\_\_\_\_

)		1	1
FREG	PZT VOLTS	POSITION	ATTENUATION
H2		MILLIVOLT	
/	±200	400	0
.2		480	+1,583
3		430	+1.583
4		465	+ 1.308
5		430	+ 0.628
6 ·		435	+0.728
7		425	+0.526
8		425	+0.526
9		415	+0.319
10		410	+0.214
20		280	- 3,098
40		280	- 3,098
60		290	- 2.793
80	·	280	-3.098
100		250	- 4.082
.120	naacija karin	205	-5.806
140	And they are End that	250	- 4.082
160		230	_4.806
180	4 (remarghant)	175	-7.180
200	History (Lit) Amer	80	-13.979
240		80.	-13,979
255 A	AK	880	+ 6.848
260		80	-13.979

•		SUBJECT	SHEET NO. 4 OF 7
I	4.2.2.3	THRESHOLD	
	.020	VOLT X.01908 50	= 0,000385E
	· ·		
I De	ete rmine	VOLTAGE/ANGLE	GAIN - AZIMUTH AXIS
1. 1	MIRROR AZ	= (+655 - 85) M IN 800 VOLTS	= 0.7125 x 10 1N VOLT
		•	

BY	_DATE	SUBJECT	SHEET NO. 5 OF 7
CHKD. BY	DATE		JOB NO

VI 4.2.2.1 TOTAL ANGULAR RANGE.

$$\frac{(793-85)^{10}(1N \times 1.0)}{7.761N} = 9.1237 \times 10^{-5} \frac{RAD}{1000}$$

Q = 91.237 x 15 RAO /1000 V 18.819 SEC 1000 VOLT 4.848 x 10 6 RAO SEC + 9.409 SEC.

641N = 0.0188 SEC VOLT

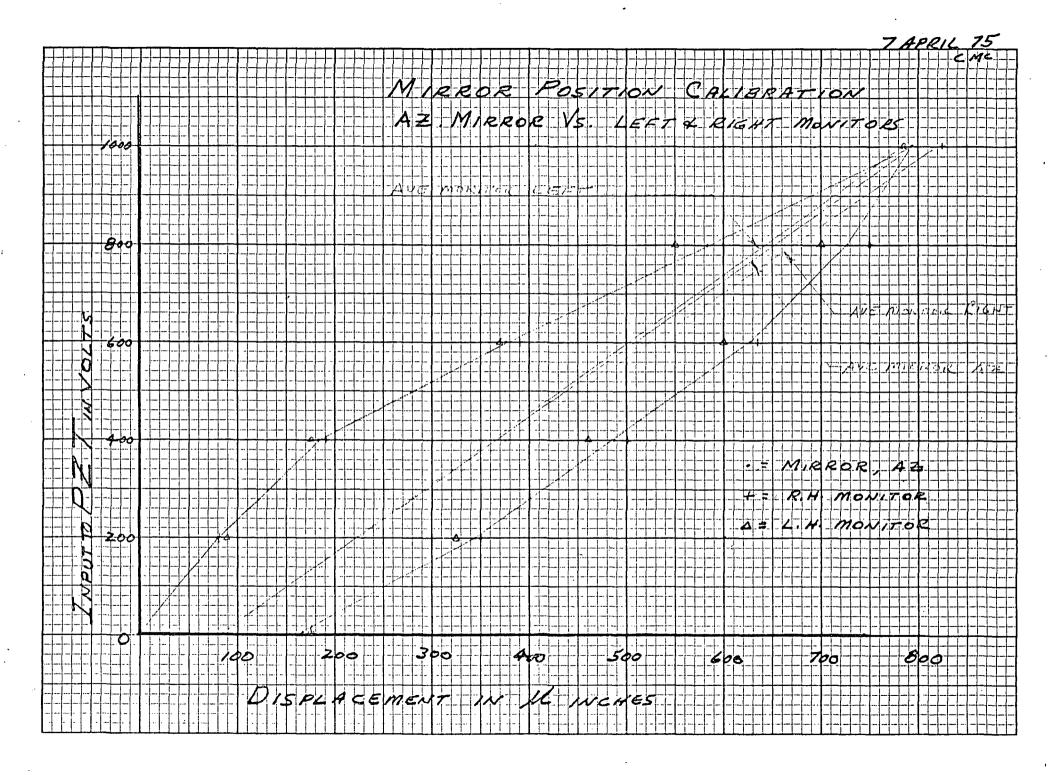
III 4.2.2.2 FREQUENCY RESPONSE @ 4 HZ

 $db = 20 \cos \frac{T_i}{T_2} = 20 \cos \frac{350}{285} = +1.784$ 

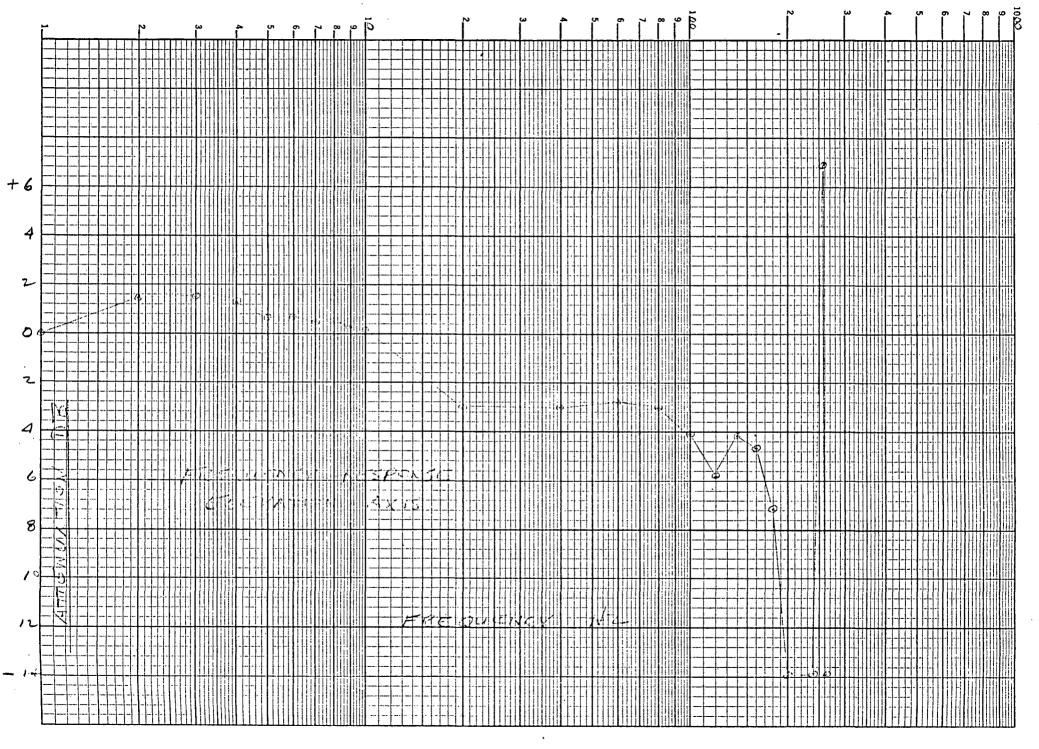
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FREQ	PZT	POSITION AZ MILLIVOLT	ATTENUATION db
1	±200	285	0
2		355	+1.908
3		355	+1.908
4		350	+1.784
5		335	+1.404
6		330	+1.273
7		320	+1,006
8		315	+.869
9	n destroya	310	+,730
10	And Andrews (Andrews	310	+.730
20	La valente de la companya de la comp	235	-1.675
40		235	-1,675
60		250	-1.138
%0		250	-1.138
100		245	-1.3/3
120		210	-2.653
140	70.	2.35	-1,675
160		335	+1,404
180		190	-3,522
190		320	
200		110	-8.269
220	·	125	-7.158
2 40		125	-7,158
260	·	410	+3.158
	·		

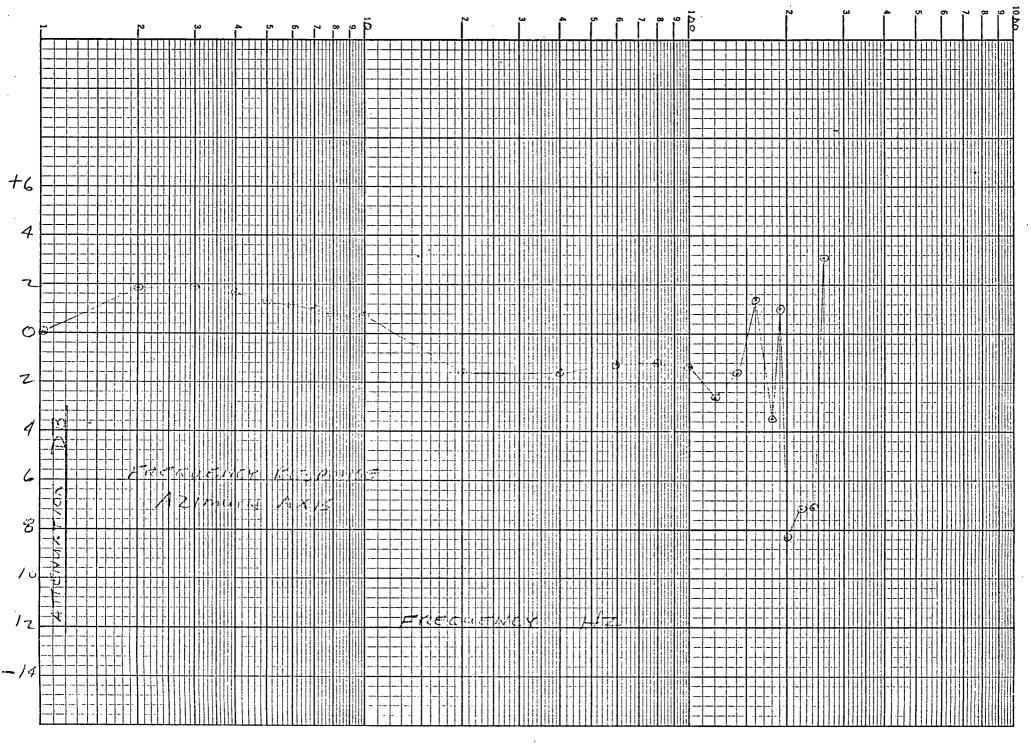
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III 4.2.2.3	THRESHOLD  VOLT X 0.0188 SEC =	0.00094 SEC



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REUFFEL & ESSER CO.



FREDUENCY AZ. AXIS 

CEEPVENCY AZ. AXIS RESPONSE 500 d 100 Spi 1320 20 40 125 20 ARAP VENCY 2

7 APRIL 75

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